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RFNC-VNIITF MULTIFUNCTIONAL SHOCK TUBE FOR INVESTIGATING THE EVOLUTION OF INSTABILITIES IN UNSTATIONARY GASDYNAMIC FLOWS

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Abstract

Parameters of flows in the RFNC-VNIITF shock tube at its operation under three modes are given. In the first, ode a stationary shock wave is formed. This makes it possible to investigate the evolution of the Richtmyer-Meshkov instability and the turbulence induced by it. In the second mode, in the shock tube a nonstationary shock wave is formed that makes it possible to investigate the behaviour of the contact boundaries of different density gases when the conditions for the evolution of the Richtmyer-Meshkov and Rayleigh-Taylor instability are realized. In the third mode a compression wave is formed that makes it possible to investigate the evolution of the Rayleigh-Taylor instability and the turbulence induced by it.

1 Introduction

In spite of a great number of investigations performed with a view to study the evolution of hydrodynamic instabilities and turbulent mixing associated with them in gases, there are many questions hat have not been studied as a result of the experimental technique imperfection.

When studying the turbulence induced by the action of the Richtmyer-Meshkov instability, a great uncertainty takes place, which is associated with the parameters of diaphragms separating the different density gases at the initial instant of time. At the same time, the influence of the diaphragms is such that the obtained turbulent flow "does not forget" the initial conditions. This leads to the turbulent flow structure distortion and, as a consequence, to great errors when measuring the turbulence parameters.

When studying the turbulence induced by the action of the Rayleigh-Taylor instability, the role of the separating diaphragms is also rather important. This problem is especially significant when studying the self-similar mode of mixing at which it is important to have the self-similar spectrum of perturbations at the contact boundary of different density gases. A great uncertainty is, possibly, associated with this circumstance in the determination of the non-dimensional rate of different density gases in the self-similar mode. This mode is characterized by the constant rate of the mixing zone width growth, at the same time, the mixing zone width **L** depends only on the density ratio of miscible media **n** or Atwood number A = (**n**—1) / (**n** + 1), the contact boundary acceleration $\mathbf{g_1}$ and time **t**:

$$L \sim A g_1 t^2. \qquad (1)$$

The self-similar mode of the gravitational turbulent mixing is everywhere used both for the calibration of the semiempirical models of mixing and for the mathematical modeling of mixing processes due to the minimum number of parameters determining this mode. The proportionality coefficient in the relation (1), which represents the non-dimensional rate of mixing, is determined in experiments and is estimated at the numerical modeling.

It is known [1] that the gravitational turbulent mixing process of different density media is processed of the definite asymmetry which consists in the fact the fronts of the penetration of the light medium into the heavy one and the heavy medium into the light one are spreading with different velocity. Historically the non-dimensional rate α_b of spreading the light medium front into the heavy one is assumed to be the characteristic of the gravitational turbulent mixing [2]. Denoting the light medium penetration front coordinate counted off from the contact boundary as L_{12}, it is possible to write down

$$L_{12} = 2\,\alpha AS, \qquad (2)$$

where $\mathbf{S} = \mathbf{gt}^2/2$.

Results obtained in experiments with different density liquids [1- 4] give the value of **a** being found in the range of

$$\alpha = 0.06 – 0.07$$

At the same time, the results obtained in the work [5] with different density gases give the magnitude of this value, which exceeds the above one shown more than by a factor of two. The reasons of such a difference have not been elucidated up to now. It may be proposed that in the work [5] either the conditions of self-similarity were not satisfied in the set-up of experiments or the measurements were made a the nonlinear stage of the Rayleigh-Taylor instability evolution, when the initial experimental conditions “were not yet forgotten”. The last argument is supported by the absence of the direct control of the initial conditions when performing experiments in this work. Moreover, the factor of compressibility can exert an influence on the result in the work [5]. However, the investigations performed in the work [6] with compressible media have shown that the values of α_b for different combinations of gases are found in the range of

$$\alpha = 0.052 – 0.098.$$

In the works [7,8] the numerical three-dimensional modeling of the gravitational turbulent mixing evolution has been carried out by means of different mathematical codes. In the work [7] the value of

$$\alpha \approx 0.052$$

was obtained, but in work [8] this value is in the range of

$$\alpha = 0.04 – 0.06.$$

Thus, it is seen that the results obtained in the work [5] for gases are contradictory. This contradiction is, most likely, associated with the experimental technique imperfection. The study of the turbulence induced by the successive action of the Richtmyer-Meshkov and Rayleigh-Taylor instabilities has not yet been performed up to now. However, this situation is rather often realized when studying the operation of laser targets in the problem of the inertial thermonuclear fusion. The absence of such work being set up under laboratory conditions is, apparently, associated with the absence of the appropriate experimental technique.

The multifunctional shock tube (MST) being developed at present in RFNC-VNIITF will make it possible to solve a number of fundamental problems of nonstationary turbulence which were described above. In the present work three modes of the MST operation associated with the shown problems are described. This development has been the result of the RFNC-VNIITF and LLNL collaboration and initially it has been known as the Project “BIZON”.

2 Multifunctional shock tube with driver I

The physical scheme of MST with driver I is presented in Fig 1.

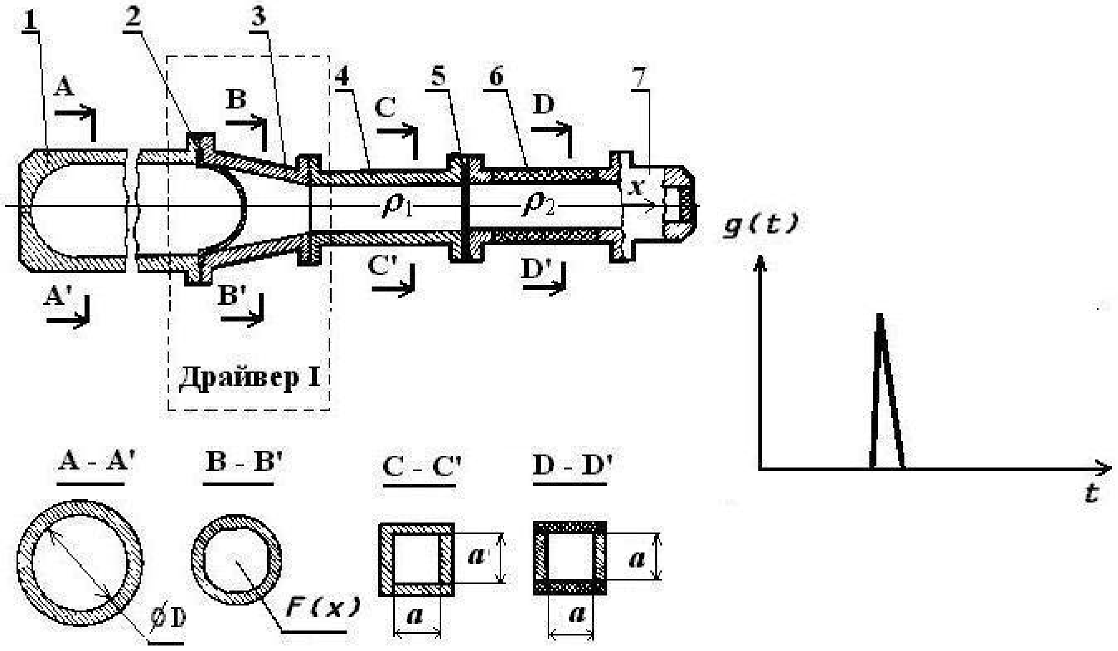

Fig. 1. The physical scheme of MST with driver I

This driver intends to be used for studying the Richtmyer-Meshkov instability and the turbulent mixing induced by it. One of the investigated gases with density ρ_1 is located in the measuring section I (4), second gas with density ρ_2 - in the measuring section II (6). At the initial instant of time a separating membrane (5) is found between gases. The composition of driver I includes a high pressure chamber (1), a high pressure membrane (2), a transitional section (3) and a part of the measuring section I. Driver I operates as follows. Gas is forced into the high-pressure chamber up to such pressure $\mathbf{P}_o$, at which the high-pressure membrane is opened. The gas flow rushes into the transitional section and then into the measuring section I creating a shock wave (SW). The function of the transitional section consists in coordinating the round cross-section **A-A** of the high-pressure chamber with the square cross-section **C-C** of the measuring section I. The cross-section **A-A** of the high-pressure chamber is chosen to be round proceeding from the considerations of its strength and technology to mount the high-pressure membrane on it. **C-C** and **D-D** cross-section of the measuring sections were chosen to be right-angled (square), proceeding from the convenience to register the turbulent mixing parameters by the light techniques. The cross-section of the other form would induce difficulties associated either with taking into account the additional refraction of light beams or with mounting the plane transparent windows on the non-planar walls of the measuring sections. The transitional section along the axis x is of a variable cross-section **F(x)** which changes from the round cross-section to the square one. At the same time, the gas flow form is smoothly changed. The intensity of the shock wave (SW) being created is determined by the value of pressure $\mathbf{P}_o$. The part of the measuring section I is used to generate a stationary SW propagating through a low-pressure gas. The required length of the stationary SW determines the length of this part of the measuring section. As a result of the SW passage through the contact boundary of gases, the contact boundary undergoes he impulsive acceleration whose character is shown in the right part of Fig.1. Mach number of the stationary SW generated by the driver I amounts to $\mathbf{M} \leq 5$.

The separating membrane (5) performs two functions. Firstly, it stabilizes the contact boundary (CB) of different density gases while preventing the mixing of gases prior to the SW arrival at CB. Secondly, by means of this membrane, at the initial instant of time, the zone of initial perturbations with specified parameters is created at the contact boundary. Such a membrane has, actually, been developed and has been named the "specter-diaphragm". Its distinctive features are its initiation from the external force and its disappearance just after the creation of the initial perturbations zone.

The length of the measuring section II is chosen depending on the problem being solved. If a single passage of SW through the zone of mixture is required, then the outlet section (7) is used for this purpose, which prevents from the creation of the reflected shock waves.

3 Multifunctional shock tube with driver II

The physical scheme of MST with driver II is presented in Fig.2.

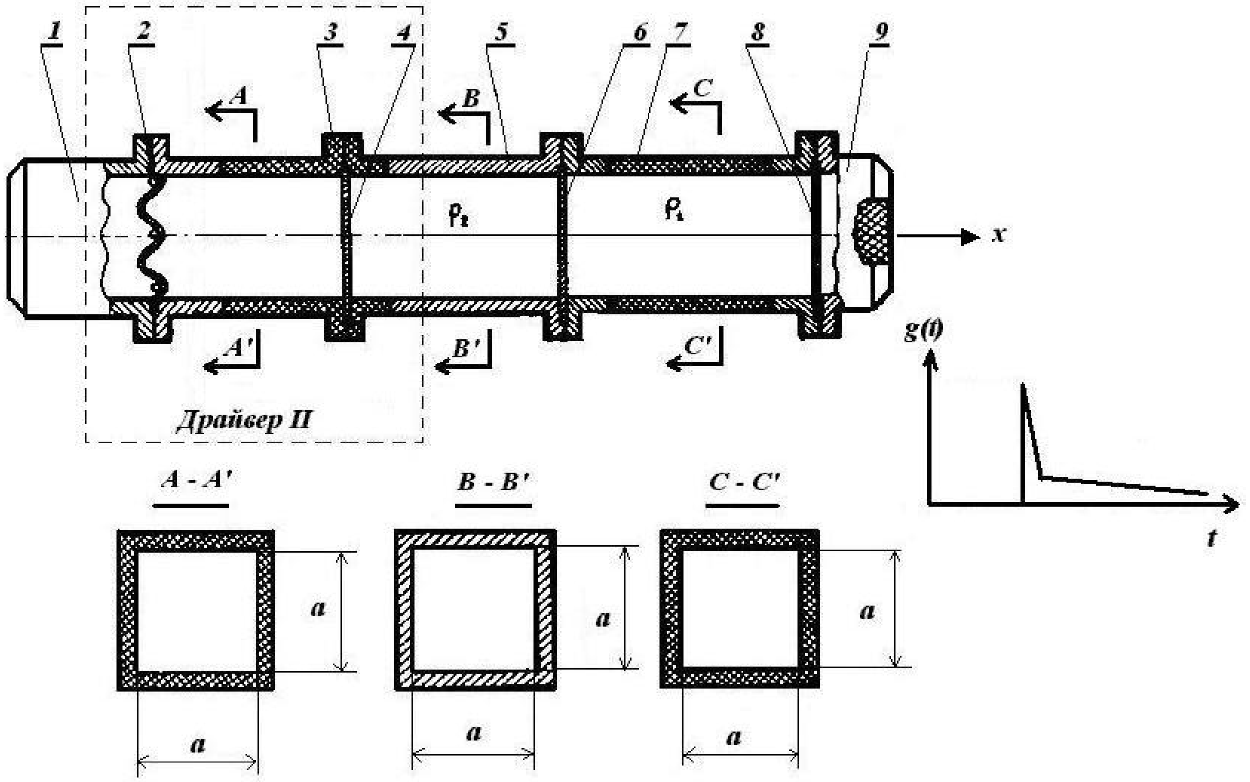

Fig.2. The physical scheme of MST with driver II

This driver intends to be used for investigating the successive action of the Richtmyer-Meshkov and Rayleigh-Taylor instabilities and the turbulent mixing induced by them. One of the gases (heavier one) being investigated and having density ρ_2 is placed into the measuring section I (5), but the lighter gas of density ρ_1 is placed into the measuring section II (7). At the initial instant of time the separating membrane (5) separates gases. This membrane performs the same function as in the case with driver I. Driver II includes a vacuum section (1), a restraining membrane (2), a section of the electrically exploded foil (EEF) (3) and the electrically exploded metal foil (4). The metal foil separates the measuring sections I from the section of EEF. The restraining membrane separates the section of EEF from the vacuum section. At the moment of the metal foil blasting a shock wave (SW) is formed which propagates into both sides. When reaching the restraining membrane (2), AW ruptures it, and gas moving behind SW begins to flow into vacuum. As a result, a rarefaction wave is formed which propagates toward the side of the measuring section I. This rarefaction wave overtakes SW, which propagate along the measuring section I. As a result, a nonstationary SW with pressure sharply dropping at the back front falls on the contact boundary. Pressure at the leading front of SW can reach $5*10^5$ Pa at the distance of 500 mm from the point of the electrically exploded foil location. Thus, the leading front of the shock wave creates the Richtmyer-Meshkov instability after passing through the contact boundary of gases and then the different density gases underwent the action of the Rayleigh-Taylor instability in the region of the contact boundary, because the gradients of pressure and density are directed in the opposite sides. For MST with the driver II the contact boundary acceleration dependence on time is shown in the right part of Fig.2. At first the contact boundary is accelerated in the pulsed mode, then it moves with almost constant acceleration. In case of MST with driver II all the section have the same square cross-section.

4 Multifunctional shock tube with driver III

The physical scheme of MST with driver III is presented in Fig.3.

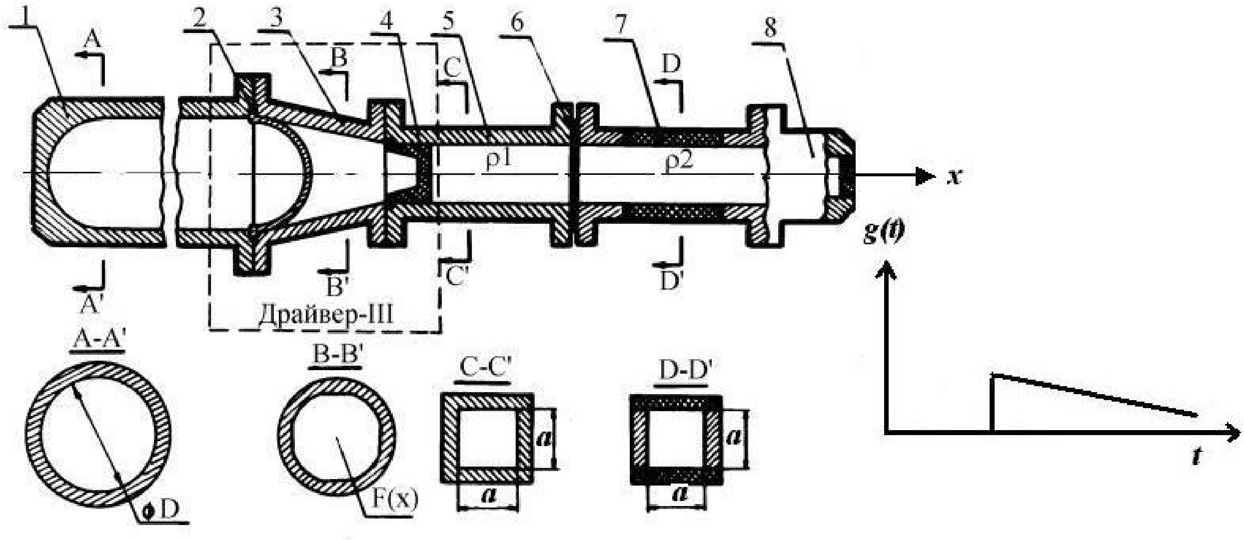

Fig.3. The physical scheme of MST with driver III

This driver intends to be used for studying the Rayleigh-Taylor instability and the turbulent mixing induced by this instability. The Lighter gas of density p_1 is placed into the measuring section I (5), the other heavier gas of density p_2 is located in the measuring section II (7). The separating membrane (6) separates gases. This membrane performs the same functions as in two preceding cases with drivers II and I. Driver III includes a high-pressure chamber (1), a high-pressure membrane (2), a transitional section (3) and a piston (4). The high-pressure chamber, the high-pressure membrane (2) and the transitional section of the driver III perform the same functions as in the driver I. However, here the gas flow does not form the stationary shock wave after passing through the transitional section, but accelerates the light piston. Then piston moving with acceleration creates a compression wave before itself, which is transformed into a shock wave in time. In the case of driver III the separating membrane and, consequently, the contact boundary of gases should be arranged at some distance from the initial position of the piston in order that by the moment of the arrival at the contact boundary the compression wave could not be transformed into SW. The pressure profile in the compression wave is such that pressure drops in the positive direction of the axis **x**. Therefore, the gradients of pressure and density in the region of the contact boundary of different density gases will be directed to the opposite sides, therefore, the contact boundary of gases will be found under condition of the Rayleigh-Taylor instability. This boundary will move along the axis **x** with almost constant acceleration whose dependence on time is shown in the right part of Fig.3. The maximum value of the contact boundary acceleration may reach the value of $g_1 = 10^5$**g**, where **g** is the acceleration of the Earth's gravitational field. The registration of the evolution process of the instability and the turbulent mixing of gases in the region of the contact boundary during its motion with acceleration is completed in the measuring section II before the piston will reach this region. Subsequently, the piston will get into the outlet section (8) where its deceleration takes place. For the same reasons which are shown when describing MST with driver I, the high pressure chamber cross-section is chosen to be round, but the cross-section of the measuring sections – to be square one. The transitional section is of the variable cross-section **F(x)** that ensures the smooth change of the form of the gas flow at its transition from the high pressure chamber to the measuring section I.

5.**Conclusion**

The multifunctional shock tube, which has been developed in RFNC-VNIITF in collaboration with colleagues from Lawrence Livermore National Laboratory and has made it possible to realize three different modes of nonstationary gas dynamic flows, will give possibility to accomplish applied tasks in the interests of solving the inertial thermonuclear fusion problems and for the development of different cumulative devices.

The multifunctional shock tube provides the following parameters of gas dynamic flows:

- in the mode with driver I a stationary shock wave with Mach number **M** ≤ 5 is generated;
- in the mode with driver II a nonstationary shock wave is generated whose initial pressure at the front is $5*10^6$ Pa with the acceleration of the contact boundary of different density gases behind the wave front $\mathbf{g}_1 \leq 10^6$ **g**, where **g** is the acceleration of the Earth's gravitational field;
- in the mode with driver III a compression wave is generated which ensures the acceleration of the contact boundaries of different density gases $\mathbf{g}_1 \leq 10^5$ **g.**

In all three modes the gases with density ration $p_2/p_1 \leq 34$ may be used.

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